

# Vector Kalman Filter

lets first extend the state model to the vector case while leaving the observation scalar.

$$\underline{s}[n] = A \underline{s}[n-1] + B u[n], n \geq 0$$

$\underline{s}$  is a  $p \times 1$  state vector

$\underline{u}$  is a  $r \times 1$  driving noise vector

$A$  is a  $p \times p$  matrix,  $B$  is a  $p \times r$  matrix (known)

$s[-1] \sim N(\mu_s, C_s)$ ,  $u[n] \sim N(0, Q)$ , ~~all indep~~

$s[-1]$  indep of  $u$

per observation  $x[n] = \underline{h}^T[n] \underline{s}[n] + w[n]$

$\underline{h}^*[n]$   $p \times 1$  vector (known)

$w[n] \sim N(0, \sigma_w^2)$  WGN indep. of  $u[n]$  and  $s[-1]$

~~Example~~ Example: stock market

State vector:  $\underline{s}[n] = \begin{bmatrix} p[n] \\ r[n] \end{bmatrix}$   $\rightarrow$  the value  
 $\rightarrow$  daily change rate

$$\underline{s}[n] = \underbrace{\begin{bmatrix} 1 & \Delta \\ 0 & 1 \end{bmatrix}}_A \underline{s}[n-1] + \begin{bmatrix} 0 \\ u[n] \end{bmatrix}$$

$\Delta = 1$  if  $n$  corresponds to day #

$$\underline{s}[-1] \sim N \left( \underbrace{\begin{pmatrix} 10000 \\ 0 \end{pmatrix}}_{\mu_s}, \underbrace{\begin{bmatrix} \sigma_p^2 & 0 \\ 0 & \sigma_r^2 \end{bmatrix}}_{C_s} \right)$$

driving noise is only in daily change rate (market sentiment)

$$B = \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix}, Q = \begin{bmatrix} 0 & 0 \\ 0 & \sigma_u^2 \end{bmatrix}$$

$u[n]$  is 0-mean with variance  $\sigma_u^2$

for instance,  $\sigma_u^2 = 200^2$

Important, probably too simplistic assumption

$$X[n] = \underbrace{\begin{bmatrix} 1 & 0 \end{bmatrix}}_{h^T} \underbrace{\begin{bmatrix} p[n] \\ r[n] \end{bmatrix}}_{s[n]} + w[n]$$

notice indep. of  $n$ .  
for this example

WGN  
with  $\sigma_w^2 = 200$

$$\hat{s}[-1|-1] = \mu_s = \begin{bmatrix} 10000 \\ 0 \end{bmatrix}$$

$$M[-1|-1] = C_s = \begin{bmatrix} 200^2 & 0 \\ 0 & 50^2 \end{bmatrix}$$

$$\text{Prediction: } \hat{s}[n|n-1] = \underbrace{\begin{bmatrix} 1 & 1 \\ 0 & 1 \end{bmatrix}}_A \hat{s}[n-1|n-1]$$

$$\text{Min. Prediction MSE: } M[n|n-1] = A M[n-1|n-1] A^T + B Q B^T$$

$$M[n|n-1] = \begin{bmatrix} 1 & 1 \\ 0 & 1 \end{bmatrix} M[n-1|n-1] \begin{bmatrix} 1 & 0 \\ 1 & 1 \end{bmatrix} + \begin{bmatrix} 0 & 0 \\ 0 & \sigma_w^2 \end{bmatrix}$$

$$\text{Kalman Gain: } K[n] = \frac{M[n|n-1] \begin{bmatrix} 1 \\ 0 \end{bmatrix}}{\sigma_w^2 + \begin{bmatrix} 1 & 0 \end{bmatrix} M[n|n-1] \begin{bmatrix} 1 \\ 0 \end{bmatrix}}$$

$$\frac{\cancel{M[n|n-1]} \begin{bmatrix} 1 \\ 0 \end{bmatrix}}{\sigma_w^2 + \begin{bmatrix} 1 & 0 \end{bmatrix} M[n|n-1] \begin{bmatrix} 1 \\ 0 \end{bmatrix}}$$

$$\text{Correction: } \hat{s}[n|n] = \hat{s}[n|n-1] + K[n] (x[n] - h^T \hat{s}[n|n-1])$$

notice scalar

$$\text{Min MSE: } M[n|n] = (I - K[n] h^T) M[n|n-1]$$

Now we will extend the observations to be vector also

$$\underline{x}[n] = \underbrace{H^T[n]}_{\substack{m \times p \\ \text{matrix}}} \underline{s}[n] + \underbrace{w[n]}_{\substack{\text{WGN} \\ N(0, C[n])}}$$

$m = \text{dim of observation vector}$   
 $p = \text{dim of state vector}$

Example = Vehicle tracking (airplane)

State vector  $\underline{s}[n] = \begin{bmatrix} p_x[n] \\ p_y[n] \\ v_x[n] \\ v_y[n] \end{bmatrix}$

$$\underline{s}[n] = \underbrace{\begin{bmatrix} 1 & 0 & \Delta & 0 \\ 0 & 1 & 0 & \Delta \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix}}_A \underline{s}[n-1] + \begin{bmatrix} 0 \\ 0 \\ u_x[n] \\ u_y[n] \end{bmatrix}$$

Constant velocity model perturbed only by small correctors & noise (wind etc.)

Driving noise only in velocity change

Take  $\Delta = 1$  (unit time intervals)

$$\underline{x}[n] = \underbrace{\begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \end{bmatrix}}_H \underline{s}[n] + \underbrace{\begin{bmatrix} w_x[n] \\ w_y[n] \end{bmatrix}}_{\text{observation noise}}$$

$m=2$  (observing only position via radar)

$$B=I, \quad Q = \begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & \sigma_{u_x}^2 & 0 \\ 0 & 0 & 0 & \sigma_{u_y}^2 \end{bmatrix}$$

Make these equal for that problem

WGN with  $C[n]$   
 $= \begin{bmatrix} \sigma_n^2 & 0 \\ 0 & \sigma_n^2 \end{bmatrix}$

$$s[-1] \sim N\left(\begin{pmatrix} 0 \\ 0 \\ 0 \end{pmatrix}, \begin{pmatrix} 10^2 & 0 & 0 \\ 0 & 10^2 & 0 \\ 0 & 0 & 10^2 \end{pmatrix} C_s\right)$$

ms is at origin & zero velocity

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$$\text{Prediction: } \hat{s}[n|n-1] = A \hat{s}[n-1|n-1]$$

$$\text{Min. Prediction MSE: } M[n|n-1] = A M[n-1|n-1] A^T + B Q B^T$$

$$\text{Kalman gain: } K[n] = M[n|n-1] H^T (C[n] + H M[n|n-1] H^T)^{-1}$$

$$\text{Correction: } \hat{s}[n|n] = \hat{s}[n|n-1] + K[n] (x[n] - H \hat{s}[n|n-1])$$

$$\text{Min MSE: } M[n|n] = (I - K[n] H) M[n|n-1]$$


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## Extended Kalman Filter

non-linear state ~~and~~ and/or observation equations

$$s[n] = a(s[n-1]) + B u[n]$$

$$x[n] = h(s[n]) + w[n]$$

Linearize  $a(s[n-1])$  about  $\hat{s}[n-1|n-1]$

Similarly linearize  $h(s[n])$

Also, particle filtering is a method to deal with this situation.